

Closed-cycle gas flow system for cooling of high T_c d.c. SQUID magnetometers

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A high T_c d.c. SQUID based magnetometer for magnetocardiography is currently under development at the University of Twente. Since such a magnetometer should be simple to use, the cooling of the system can be realized most practically by means of a cryocooler. A closed-cycle gas flow cooling system incorporating such a cooler has been designed, constructed and tested. The aimed resolution of the magnetometer is $0.1 \text{ pT Hz}^{-1/2}$. The required operating temperature for the SQUIDs is 30 to about 77 K with a stability of $2 \times 10^{-4} \text{ K Hz}^{-1/2}$. After a cool-down time of 1–2 h, a stationary cooling power of at least 0.2 W is required. In the design, helium gas is cooled by a Leybold Heraeus RG 210 cryocooler, transported through a gas line, and subsequently passed through a heat exchanger on which SQUIDs can be installed. The lowest obtainable SQUID heat exchanger temperature is $31 \pm 2 \text{ K}$. This can be reached in roughly 2–3 h with an optimal mass flow with respect to the cooling power of $6 \times 10^{-6} \text{ kg s}^{-1}$. At this mass flow the cooling power at the SQUID heat exchanger is 0.2 W at 42 K and roughly 1.2 W at 77 K. A temperature stability of 0.05 K was measured at a SQUID heat exchanger temperature of 54 K and a mass flow of $3 \times 10^{-5} \text{ kg s}^{-1}$. The experience gained with this large cooling system will be used in the design of a smaller configuration cooling system, incorporating miniature Stirling cryocoolers. In this paper the design and the construction of the present closed-cycle system are described and test results are presented.

Keywords: high T_c d.c. SQUID magnetometer; closed-cycle gas flow cooling; heat transfer

Superconducting quantum interference devices (SQUIDs) are the most sensitive magnetic flux-to-voltage converting sensors. This has opened the way for biomagnetic research by means of multichannel low T_c d.c. SQUID based magnetometer systems^{1–3}. These systems are usually cooled by liquid helium and operated in magnetically shielded rooms to obtain an extremely low-noise environment. However, they are expensive, require helium refills and cannot be transported in a simple manner. Since high T_c d.c. SQUIDs can operate at much higher temperatures (up to liquid nitrogen temperature)^{4–7}, small scale cryocoolers⁸ can be applied to cool a simple to use high T_c d.c. SQUID based magnetometer system. Reliable, small scale, turn-key cryocoolers for infrared applications are available⁹. In this paper, a closed-cycle gas flow system for cooling a high T_c d.c. SQUID based magnetometer for heart measurements is described. Although the cooling power of the Leybold Heraeus RG 210 cryocooler in the present system is oversized, experience with this large scale system can be used in the design of smaller systems, based on one or more miniature Stirling cryocoolers. Topics such as magnetic noise suppression via active or passive compensation and

elimination of cooler vibrations are under study, but beyond the scope of this paper.

In the following section of this paper, the requirements for a closed-cycle gas flow system for cooling a high T_c d.c. SQUID based magnetometer are considered. After that, the design aspects of our system are described. Experimental results are presented and discussed in the last section.

Requirements

The aim of the high T_c d.c. SQUID magnetometer used in this study is to measure adult human heart signals with a resolution of $0.1 \text{ pT Hz}^{-1/2}$ in a bandwidth of 0.1–40 Hz in the pick-up coil. This value is sufficient for heart signal measurements and can be achieved using recently fabricated high T_c d.c. SQUIDs^{4,5}. The technical requirements which are important for the magnetometer cooling system are listed in *Table 1*.

Table 1 Required features of high T_c d.c. SQUID magnetometer cooling system with an aimed resolution of $0.1 \text{ pT Hz}^{-1/2}$

Parameter	Requirement	Comment
Typical operating temperature (K)	30–77	1
Temperature stability ($\text{K Hz}^{-1/2}$)	$< 2 \times 10^{-4}$	2
Mass to be cooled (kg)	0.002	3
Cooling power (W)	≥ 0.2	3
Cool-down time (h)	1–2	3
Mean time between:		
Routine maintenance (yr)	> 0.5	4
Failure (yr)	$> 1-2$	4
Magnetic interference ($\text{T Hz}^{-1/2}$)	$\leq 10^{-13}$	5
Vibration level ($\text{degree Hz}^{-1/2}$)	$\leq 10^{-7}$	6

Comments on Table 1

- 1 The maximum operating temperature of the d.c. SQUID is limited by the intrinsic noise of presently available high T_c d.c. SQUIDs⁴⁻⁷ relative to the desired resolution. Small scale cryocoolers make operation at temperatures down to $\approx 30 \text{ K}$ possible.
- 2 One may expect that the requirement for temperature stability is related to the thermal fluctuations in the critical currents of the Josephson junctions in the d.c. SQUID, which, at low temperatures, are the main source of intrinsic noise in a high T_c d.c. SQUID. However, modulation techniques can be applied to eliminate the influence of these critical current fluctuations¹⁰. Therefore, a more stringent requirement results from the thermal expansion of the magnetometer sensing coil. Because of the small thickness of the sensing coil, the thermal expansion is mainly determined by the SrTiO_3 substrate ($\approx 5 \times 10^{-6} \text{ K}^{-1}$ at 50 K ¹¹). The required temperature stability is determined by calculating the temperature effect on the sensing coil in the earth's magnetic field (EMF) in the worst case that the EMF is perpendicular to the area of the sensing coil. A compensation coil may relax the required temperature stability in the EMF with a factor equivalent to the compensation factor for d.c. magnetic fields of the coil; a factor of roughly 10^3 has been reported¹².
- 3 A d.c. SQUID is nearly non-dissipative. Therefore, the cooling power is used mainly to compensate the stationary thermal load, which is predominantly caused by heat conduction from the supporting material and the leads to the sensor. Because the heat capacity of the sensor is small compared to the capacity of the rest of the system, the latter will determine the cool-down time.
- 4 The mean times between failure and routine maintenance are based on the situation outside cryogenic laboratories.
- 5 Depending on the uniformity of the magnetic noise due to the cryogenic system, gradiometer-type construction of the sensing coil can increase the acceptable noise level by a factor of $\approx 10^3$, a value which is determined by the common mode rejection ratio (CMRR)¹³.
- 6 The largest noise contribution arises from axial tiltings of the sensing coil in an external field. A sensing coil vibrating in the EMF results in a noise contribution of

$\approx 10^{-6} \text{ T}$ per degree. Again, a gradiometer-type configuration may increase the acceptable mechanical vibration level in the EMF by a factor equivalent to the CMRR (roughly 10^3). Moreover, a compensation coil as mentioned in comment 2 may also relax the requirement for the vibration level.

Design

Figure 1 schematically shows our cooling system. Helium gas is forced to flow in a closed cycle by means of a gas-flow pump and is cooled by an LH-RG 210 cryocooler. The cold gas is transported through a coaxial gas line, and cools a d.c. SQUID via a heat exchanger. (Hereafter a heat exchanger is denoted as hx.) The return gas is used to thermally insulate the supply gas. Because the gas-flow pump operates at room temperature, a counterflow heat exchanger (henceforth denoted as cfhx) has been incorporated. The whole closed-cycle system is vacuum insulated. A photograph of the cooling system is presented in Figure 2. The separate elements of the system will be considered in the following sections, and are described in more detail elsewhere¹⁴.

Gas flow controller

The helium gas flow in the closed-cycle system is controlled by a gas flow controller, which consists of two parallel mass flow controllers for two ranges: $4 \times 10^{-7} - 4 \times 10^{-6} \text{ kg s}^{-1}$ and $4 \times 10^{-6} - 4 \times 10^{-5} \text{ kg s}^{-1}$. The gas is pumped with an oil-free diaphragm pump to prevent oil pollution of the helium gas in the closed-cycle system. This pump is shunted by an adjustable valve to control the pressure in order to prevent overheating of the pump. The mass flow can be blocked with a valve in the flush and fill section of the gas flow controller. By means of this section the system can be cleaned and (re)filled. Pressure meters are installed at the inlet and the outlet sides of the controller. Finally, two buffer volumes (1120 and 160 cm^3) have been incorporated, which reduce the filling pressure of the system. A small overpressure of $\approx 0.1 \text{ bar}^*$ during cool-down is chosen in order to prevent pollution of the gas circuit via small leaks. The buffers also reduce the gas flow fluctuations caused by the pump, which may disturb the mass flow measurement.

The temperatures in the cooling system depend on both the mass flow \dot{m} of the helium gas and the refrigerating characteristics of the cryocooler. The maximum mass flow depends directly on the pump characteristic (pressure drop versus \dot{m}). It also depends indirectly on the temperature T , since the pressure drop in the closed-cycle system Δp_s is related to T . Based on the cryocooler refrigerating power, the measured pump characteristic and an estimate of Δp_s , a maximum mass flow \dot{m}_{max} of $2 \times 10^{-5} \text{ kg s}^{-1}$ at the start of a cool-down of the system seems possible and sufficient¹⁴. The estimate of Δp_s is based on a laminar, frictionless helium gas flow at room temperature through a transfer line with a hydraulic diameter of 1.6 mm and a length of 10 m . At lower T , Δp_s decreases.

At \dot{m}_{max} , the cool-down time of the complete system is,

* $1 \text{ bar} = 10^5 \text{ N m}^{-2}$

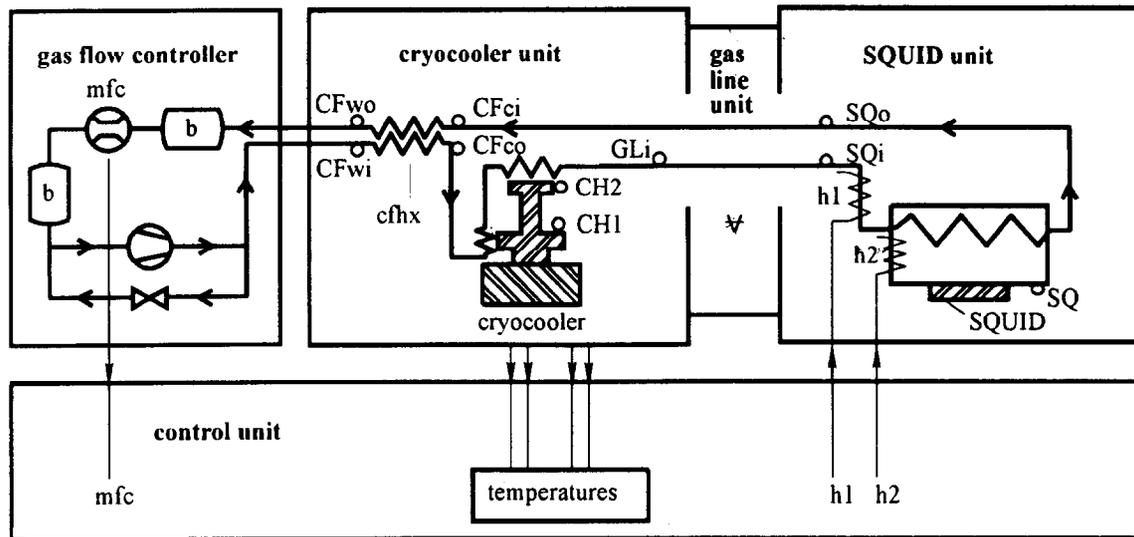


Figure 1 Scheme of the designed closed-cycle gas flow system. In this system the temperatures are measured at the first stage of the cryocooler (denoted by CH1), at the second stage of the cryocooler (CH2), at the gas supply line inlet (GLi), at the SQUID unit inlet (SQi), at the SQUID heat exchanger (SQ), at the SQUID unit outlet (SQo), at the cold inlet and cold outlet of the counterflow heat exchanger (CFci and CFco, respectively), and finally at the warm inlet and warm outlet of the counterflow heat exchanger (CFwi and CFwo, respectively). The mass flow controller is indicated by mfc, the counter flow heat exchanger by cfhx, a buffer and a heater by b and h, respectively, and vacuum by ∅

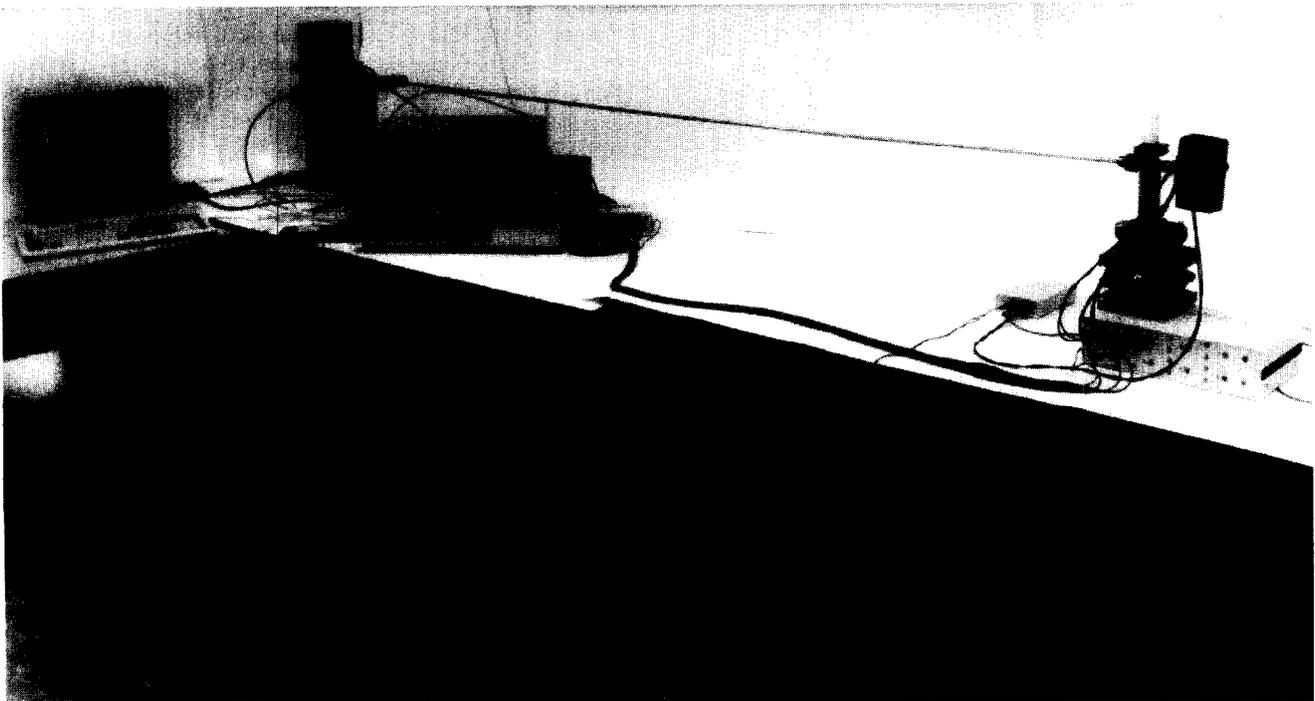


Figure 2 Photograph of the closed-cycle gas flow cooling system

for a large part, determined by the heat capacity in the cryocooler unit (cryocooler plus copper hxs). At this rate of flow, a cool-down time of ≈ 2 h is expected. Cooling the cryocooler plus the hxs separately would take ≈ 1 h (reference 15).

Cryocooler unit

The cryocooler unit utilizes an LH-RG 210 cryocooler and a cfhx, both placed in a vacuum environment. The Gifford-McMahon type cryocooler consists of a compressor and a two-stage cold head. It has a refrigeration capacity of 14 W at 80 K on the first stage combined with 2 W at 20 K on

the second stage¹⁵. Two copper hxs are mounted on the two stages and are connected by a flexible spiral-shaped stainless steel tube (inner and outer diameters of 1.6 and 2 mm, respectively). These hxs are fabricated from copper blocks in which grooves have been machined to serve as gas channels. The cross-sectional area of these square channels is ≈ 2.3 mm² and the lengths of the first and second stage hxs are 0.7 and 1 m, respectively. The hx on the second stage is surrounded by a copper radiation shield, which has been connected to the first stage. The cfhx keeps the temperature difference between the environment and the first cold head stage more or less constant. It consists of two coaxial stainless steel tubes with a length of 1 m. The inner tube, which



Figure 3 Schematic cross-section of the gas line

is connected to the return gas transfer line, has inner and outer diameters of 1.6 and 2 mm, respectively. The outer tube has diameters of 3 and 4 mm, respectively. Both the inner side of the cryocooler unit and the copper radiation shield are covered by multilayer insulation (MLI).

Gas line unit

Measurements of the magnetic interference originating from the cryocooler were performed to determine the required distance between the cryocooler and the SQUID. A large magnetic noise contribution ($> 0.1 \text{ pT Hz}^{-1/2}$) from the cryocooler in the frequency bandwidth relevant to cardiomagnetometry (0.1–40 Hz) should be avoided. Because the compressor can be located at $\approx 2 \text{ m}$ from the cold head, the main noise contribution at the SQUID appeared to be due to the cold head. Based on the magnetic noise measurements, a separation of 2.5 m between the cold head and the sensor was chosen¹⁴. This distance may be reduced if the cold head is shielded with mu-metal. However, problems might arise with respect to vibrations of the magnetic shield.

The cold helium gas is transported through the 2.5 m long coaxial gas transfer line, in which both the supply and return lines are combined (see *Figure 3*). Because a flexible coaxial line is expensive and, as expected, not necessary in future small scale configurations, the transfer line is fabricated in a rigid manner. All tubes are made of stainless steel. The inner (and outer) diameters of the four tubes are, respectively, 1.6 (2), 3 (4), 5 (6) and 7 (8) mm. The cold gas from the cryocooler unit flows through the inner tube, which is thermally insulated by a vacuum space around this tube and the return gas line. The return gas line is only insulated by a vacuum space. Nylon wire is used as spacer material in both vacuum spaces to avoid direct contact between adjacent lines. At the sensor end of the gas line, the supply and return lines split into two separate lines (see *Figure 4a*). The stainless steel transfer lines are fixed with

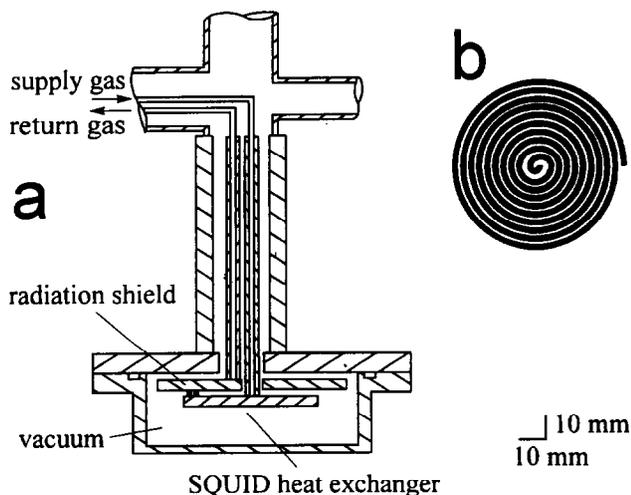


Figure 4 Schematic views of (a) the SQUID unit and (b) the spiral channel structure of the heat exchangers (the spiral of the radiation shield is depicted)

nylon wires to avoid damage to the SQUID unit due to thermal shrinkage and expansion of the gas line elements.

SQUID unit

The SQUID unit is shown schematically in *Figure 4a*. The cold gas from the gas line unit flows directly into the lower hx. In this way, a d.c. SQUID mounted on the bottom side of this SQUID hx can be cooled. The helium gas returns to the gas line unit via a second hx which serves as a radiation shield.

To reduce the thermal noise and to prevent magnetic noise from metals in the vicinity of the d.c. SQUID, the SQUID unit was made of glass-epoxy G11 (HGW 2372.4)¹⁴. This material is easily machinable with standard tools and we have considerable cryogenic experience of it. At the entrance of the SQUID unit the stainless steel transfer lines have been glued to two small glass-epoxy tubes. The small tubes were glued to the hx's at the bottom and are wrapped in MLI. The tubes have inner and outer diameters of 2 and 5 mm, respectively, with a length of $\approx 0.12 \text{ m}$.

The SQUID hx and the radiation shield have a flat spiral channel structure (see *Figure 4b*) to limit the temperature gradient over the plates. The plates are 3.9 mm thick and have diameters of 5 and 7 cm, respectively. The machined channels are 2 mm wide, 1.5 mm deep and have lengths of 0.6 and 0.9 m. Glass-epoxy spacers (3 mm long) have been glued between the SQUID hx and the radiation shield, and also between the radiation shield and the top plate of the SQUID box. The hx's are situated in a vacuum environment. At the bottom side of the SQUID hx several (≈ 20) layers of MLI have been placed.

Two heaters made of high-ohmic wire are fixed in this unit. The first one has been wound around the stainless steel tube at the top of the SQUID unit, just before the transition of the supply gas line to the glass-epoxy tube. This heater can be used to control the temperature stability at the SQUID hx. For measuring the cooling power at the SQUID hx, a second heater was fixed in the middle of the bottom side of this hx.

Control unit

The temperatures in the system are measured with Lake Shore DT-470/471 silicon temperature diodes. The estimated error in T is 1.4 K for $T < 100 \text{ K}$ and $(0.6 + 0.008T) \text{ K}$ for $100 \text{ K} < T < 300 \text{ K}$. The temperatures at the first and second stage of the cold head are measured by diodes that have been installed in the copper hx's mounted on these stages. In the SQUID unit the diode has been fixed to the plate by aluminium tape. All the other diodes are fixed to the gas transfer line with an aluminium oxide holder to establish good thermal contact with no electrical leakage currents. The aluminium oxide holder consists of two pieces, which are pressed together by a small aluminium tube or, in the case of the diodes for measuring the inlet and outlet temperature of the SQUID unit, by aluminium tape. Apiezon grease is used to increase thermal contact between the different parts. The measured temperatures as well as the measured mass flow are automatically and on-line registered by the control unit. The mass flow is controlled by two mass flow controllers. The estimated error in \dot{m} is 1–2%. Both heaters in the SQUID unit can be fed with separate sources from the control unit.

Table 2 Estimated pressure and temperature drops, Δp and ΔT , respectively, for system units at initial and final thermodynamic states

Unit	Volume (10^{-6} m^3)	$\dot{m} = 2 \times 10^{-5} \text{ kg s}^{-1}$	$\dot{m} = 1 \times 10^{-6} \text{ kg s}^{-1}$ (steady state)	
		Δp (10^5 Pa)	Δp (10^2 Pa)	ΔT (K)
Gas flow controller	1350	-0.001	-0.04	0
Counterflow hx-supply	6	-0.27	-3.7	-(50-100)
Cold head: first and second stage	4	-0.29	-0.4	-200
Supply gas line	6	-0.43	-0.5	20
SQUID unit	3	-0.35	-0.4	15
Return gas line	21	-0.29	-0.7	165
Counterflow hx-return	3	-0.22	-1.9	50-100
Total system (without pump)	1393	-1.85	-7.6	-

Estimated pressure and temperature drops

Based on the hydraulic diameters in the system, the characteristics of the cooler and the pump, and some assumptions on conductive and radiative heat flows, the temperature and pressure drops, ΔT and Δp , have been evaluated. In Table 2 the Δp values are presented for the warm start-up situation, whereas for the cold steady state both Δp and ΔT values are given. The higher \dot{m} is supposed to be used at the start of cool-down, resulting in the highest Δp values. The lower \dot{m} will be used in the cold steady state.

Results

Figure 5 shows a typical cool-down recorded by the 10 temperature diodes in the system. It was measured at a constant \dot{m} of $1.4 \times 10^{-5} \text{ kg s}^{-1}$ and it shows a stationary SQUID temperature T_{SQ} of 33 K after ≈ 2 h. This T_{SQ} was measured in the centre of the bottom side of the SQUID hx. The figure shows cool-down of the system, starting from the first stage temperature of the cryocooler (T_{CH1}), followed by the second stage temperature of the cryocooler (T_{CH2}), the gas supply line inlet temperature (T_{GLi}), the SQUID inlet temperature (T_{SQi}), T_{SQ} and the SQUID outlet temperature (T_{SQo}). Later on the cfhx-cold in and -cold out temperatures (T_{CFci} and T_{CFco} , respectively) follow. As expected the cfhx-warm in and -warm out temperatures (T_{CFwi} and T_{CFwo} , respectively) remain almost constant at room temperature. T_{CH1} cools down exponentially for about 30 min. Subsequently, when the cfhx becomes effective, T_{CH1} decreases linearly with time for more than 30 min,

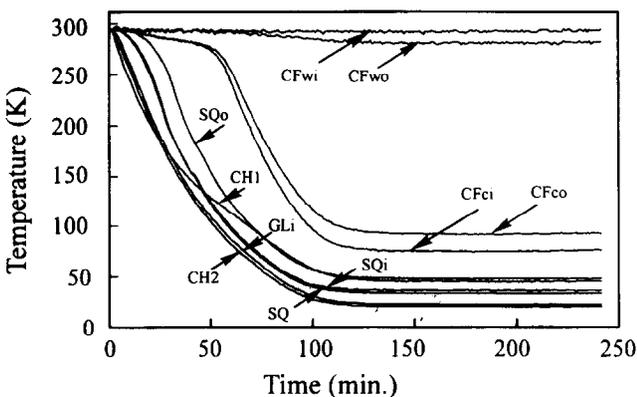


Figure 5 Cool-down recorded by the 10 temperature diodes in the system at constant $\dot{m} = 1.4 \times 10^{-5} \text{ kg s}^{-1}$. The different positions in the system are denoted by abbreviations as defined in the caption to Figure 1

followed by a second exponential cool-down tail. T_{CH2} appears to cool down continuously to much lower temperatures in an exponential way.

After almost 1 h, T_{CFci} and T_{CFco} start to cool down sharply, showing that the system including the total gas line has been cooled down up to the cfhx. This also indicates good thermal insulation between the supply and return gas line. In general, at constant cooling power the time between two sharp temperature decreases as seen in Figure 5 can be considered an indication of the heat capacity of the material between the two temperature-measuring points. The mass of the material between the positions of the diodes measuring T_{GLi} and T_{SQi} is roughly a factor of 10 smaller than the mass of the material between the positions of the diodes measuring T_{SQo} and T_{CFci} . As expected, the figure shows that the ratio of the times between the sharp drops of T_{GLi} and T_{SQi} , and T_{SQo} and T_{CFci} is also roughly 10.

The temperature stability was measured with the diode ≈ 3 cm from the centre of the bottom side of the SQUID hx (T'_{SQ}). Figure 6 shows the stability of T'_{SQ} as a function of time. In this experiment, \dot{m} was kept constant at $2 \times 10^{-5} \text{ kg s}^{-1}$ and T'_{SQ} was 40 K but not completely stabilized. Inset A in this figure shows a temperature stability of

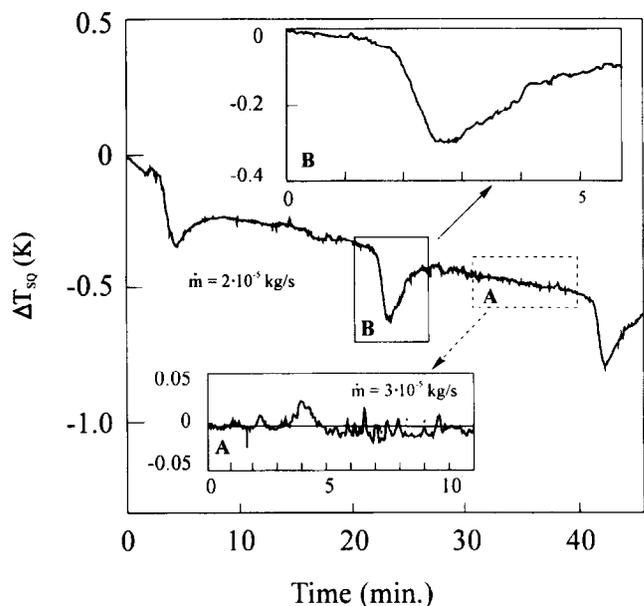


Figure 6 Temperature stability of T'_{SQ} (denoted by ΔT_{SQ}) as a function of time measured on the bottom side of the SQUID heat exchanger at $\dot{m} = 2 \times 10^{-5} \text{ kg s}^{-1}$ and $T'_{SQ} = 40 \text{ K}$ (not stabilized). Inset A was measured at $\dot{m} = 3 \times 10^{-5} \text{ kg s}^{-1}$ and $T'_{SQ} = 54 \text{ K}$ (steady state)

≈ 0.05 K measured for ≈ 12 min at a flow of 3×10^{-5} kg s $^{-1}$ and a steady state T'_{SQ} of 54 K. The long term stability is limited by small, frequently occurring temperature dips in T'_{SQ} (maximum 0.4 K, duration 3–4 min, approximately once per 18 min; see inset B). Such dips are also observed in T_{CH1} and T_{CH2} , although the maximum depth of the dips in T_{CH1} is smaller. These dips are most probably caused by temporary blockings of the bypass from the oil separator to the pump in the compressor of the cryocooler due to oil droplets that flow back through the bypass to the pump, causing pressure rises and consequently a lower temperature at the cryocooler stages. The drift effect in the figure (estimated at roughly -0.8 K h $^{-1}$) was mainly due to the fact that the system was not completely stabilized. The small dips and their delayed influence might also partly have determined the drift.

The long term variations do not have to be reduced because they lie outside the frequency range of interest. The temperature fluctuation of 0.05 K at the SQUID hx, however, is relevant and is roughly a factor 40 higher than the required temperature stability. A temperature control system via regulation of \dot{m} is under investigation. Also the requirement with respect to temperature stability can be relaxed when the EMF is compensated by means of a coil.

Figure 7 shows the 10 different temperatures at steady state as a function of \dot{m} . A minimum T_{SQ} of 31 K is found at 6×10^{-6} kg s $^{-1}$ (henceforth denoted as the optimal mass flow with respect to the cooling power \dot{m}^*). For the lowest \dot{m} in the figure the error in T is larger than the error given above, because the system was not completely stationary. The reason why T_{SQi} exceeds T_{SQ} for several \dot{m} values is not clear. Possibly the thermal contact between the diode measuring T_{SQ} and the glass-epoxy SQUID hx on which the diode had been fixed was better than the contact between the diode measuring the gas temperature T_{SQi} inside a stainless steel tube via an aluminium oxide holder.

The steady state temperatures of the system at a certain \dot{m} value are determined by the balance between heat supply

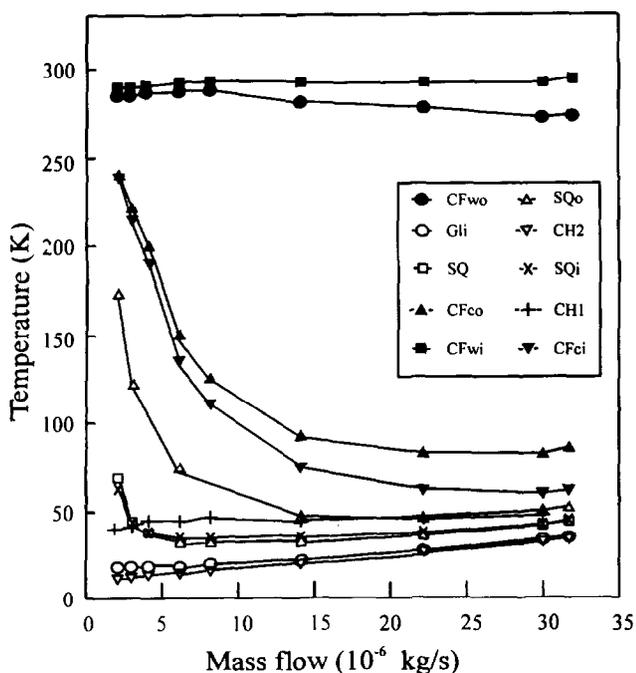


Figure 7 Steady state temperatures of the cooling system as a function of \dot{m} . The abbreviations indicate the different positions in the system and are defined in the caption to Figure 1

and heat loss in the different units of the cooling system. The factors involved are: 1, the cooling power supplied by the cryocooler; 2, heat leakage into the cryocooler, gas line and SQUID unit; and 3, the heat flows in the cfhx determining its efficiency. The stationary heat loss in the supply gas line can be defined as $\dot{q}_{GLs} = \dot{m}c_p(T_{SQi} - T_{GLi})$, the loss in the SQUID unit as $\dot{q}_{SQ} = \dot{m}c_p(T_{SQo} - T_{SQi})$, that in the return gas line as $\dot{q}_{GLr} = \dot{m}c_p(T_{CFci} - T_{SQo})$ and that in the cfhx as $\dot{q}_{cfhx} = \dot{m}c_p(T_{CFco} - T_{CFci})$, where c_p is the helium heat capacity. These losses are shown as a function of \dot{m} in Figure 8. The stationary energy balance is given by: $\dot{q}_{cc,eff} + \dot{q}_{GLs} + \dot{q}_{SQ} + \dot{q}_{GLr} + \dot{q}_{cfhx} = 0$, in which the effective cryocooler cooling power $\dot{q}_{cc,eff}$ is defined as $\dot{q}_{cc,eff} = \dot{q}_{cc} + \dot{q}_{cc,loss} = \dot{m}c_p(T_{GLi} - T_{CFco})$; \dot{q}_{cc} is the value of the cryocooler power, which is a function of T_{CH1} and T_{CH2} and can be evaluated from the specifications¹⁵. The difference $\dot{q}_{cc,loss}$ is due to conductive and radiative losses in the cryocooler. It can also partly be caused by the fact that the real cryocooler power deviates from the specified cryocooler power due to ageing (11 years) and/or a too low filling pressure of the compressor during the measurements.

In Figure 8 the absolute values of $\dot{q}_{cc,eff}$ and \dot{q}_{cc} are given as well; the errors in $\dot{q}_{cc,eff}$ and in the heat losses are roughly of the order of the marker size (at high \dot{m}) or smaller (at low \dot{m}). The estimated errors in \dot{q}_{cc} due to the limited accuracy in reading the specification curves are indicated by error bars; \dot{q}_{cc} for the two highest \dot{m} values could not be derived from the specifications.

The efficiency of the cfhx is determined by the heat transfer coefficient (inclusive properties of the helium gas: c_p , the thermal conductivity and the dynamic viscosity), the mass flow, and the tube geometry (the length and diameters), if two assumptions are satisfied. The first is that there is no heat flow from outside to the cfhx, which can be established by covering the tubes of the cfhx with MLI. The second is that the heat flow due to conduction down the walls of the tubes of the cfhx is negligible. The efficiency of an ideal cfhx tends to 1 for \dot{m} close to zero, and levels off for increasing \dot{m} , approaching zero for infinite \dot{m} (reference 16).

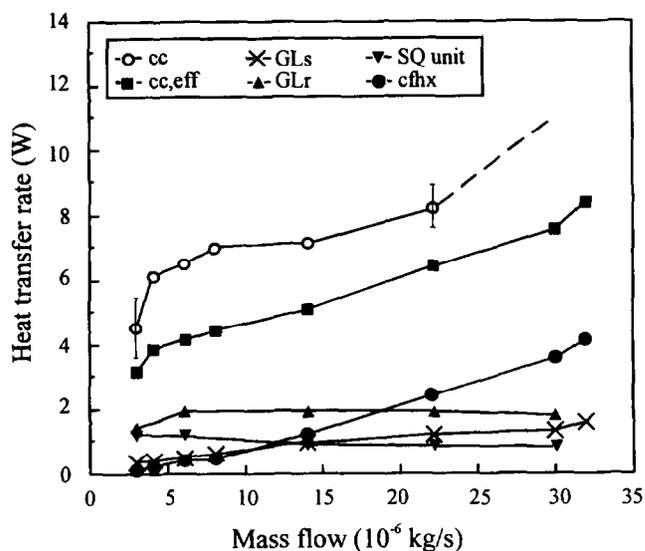


Figure 8 Steady state heat loss in the supply gas line (\dot{q}_{GLs} , indicated by its subscript only), return gas line (\dot{q}_{GLr}), SQUID unit (\dot{q}_{SQ}) and counterflow heat exchanger (\dot{q}_{cfhx}), and the absolute values of the (effective) cryocooler cooling power (\dot{q}_{cc} and $\dot{q}_{cc,eff}$) as a function of \dot{m}

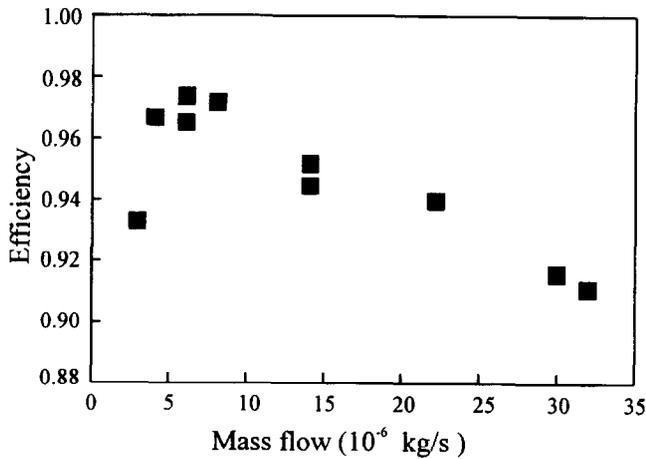


Figure 9 Efficiency η of the counterflow heat exchanger as a function of \dot{m} . Errors at low \dot{m} are $\approx 4\%$ and at high \dot{m} are $\approx 3\%$

The efficiency of the cfhx is defined as $\eta = \Delta_1/\Delta_2$, in which $\Delta_1 = T_{CFwo} - T_{CFci}$ and $\Delta_2 = T_{CFwi} - T_{CFci}$ (reference 16). Figure 8 shows an increase in \dot{q}_{cfhx} with \dot{m} at higher \dot{m} values. This non-ideal character of the cfhx can also be seen in Figure 7: it shows that both Δ_1 and Δ_2 increase with \dot{m} , but Δ_2 increases relatively faster than Δ_1 for $\dot{m} > \dot{m}^*$, resulting in a decrease in η . A comparison between T_{CFwi} and T_{CFwo} also demonstrates the non-ideal character: T_{CFwi} is almost constant for all \dot{m} , whereas T_{CFwo} decreases slightly for $\dot{m} > \dot{m}^*$. Figure 9 shows η as a function of \dot{m} . The errors are estimated at roughly 4% at low \dot{m} and 3% at high \dot{m} . The figure shows a tendency for decreasing η with increasing \dot{m} .

The cfhx is important in order to obtain a low T_{SQ} value. The lowest T_{SQ} is reached for $\dot{m} = \dot{m}^*$ at which: 1, T_{CH2} is relatively low; 2, $\dot{q}_{cc,eff}$ is relatively large compared with the leakage into the gas line and SQUID unit; and 3, the efficiency of the cfhx is relatively good. For $\dot{m} > \dot{m}^*$ the temperature difference between T_{CH2} and T_{SQ} decreases with increasing \dot{m} , because $\dot{q}_{cc,eff}$ becomes relatively larger in comparison with the heat leakages into the gas line and SQUID unit with increasing \dot{m} .

Figure 10 shows the cooling power \dot{q}_{sensor} at the SQUID hx as a function of T'_{SQ} for different \dot{m} values. In this experiment the heater was fixed at the centre on the bottom side of the SQUID hx. In this figure the values of $\dot{q}_{cc,eff}$ are given which correspond to the measured \dot{q}_{sensor} values, indicating the efficiency of \dot{q}_{sensor} with respect to $\dot{q}_{cc,eff}$. The $\dot{q}_{cc,eff}$ values without brackets correspond to the steady

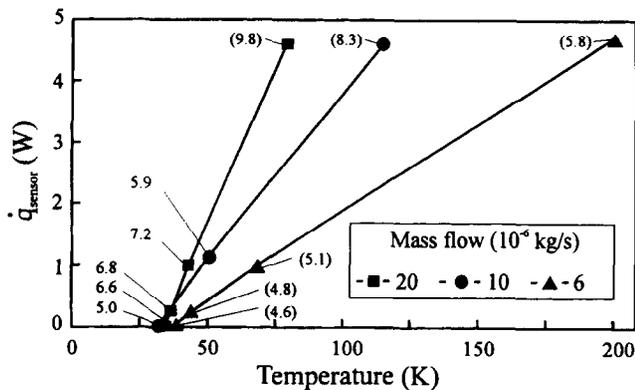


Figure 10 Cooling power \dot{q}_{sensor} at the SQUID heat exchanger as a function of T'_{SQ} for different \dot{m} values; the given values corresponding to the measured points are $\dot{q}_{cc,eff}$ values

states, whereas the values between brackets are not fully stabilized (resulting errors are roughly a few per cent). Because the temperature diode was at different positions for measuring T_{SQ} and T'_{SQ} , the T'_{SQ} values at $\dot{q}_{sensor} = 0$ W in this figure differ in comparison with the T_{SQ} values in Figure 7. The slope of \dot{q}_{sensor} versus T'_{SQ} is not proportional to \dot{m} , because a variation in \dot{m} affects the total system and its temperature distribution.

Conclusions

A closed-cycle gas flow system for cooling a high T_c d.c. SQUID based magnetometer by means of a cryocooler has been designed, constructed and tested. The cool-down curves and the steady state temperatures are measured at 10 positions in the cooling system for different mass flows. The lowest attainable steady state SQUID temperature T_{SQ} is 31 ± 2 K. It can be reached in roughly 2–3 h with an optimal mass flow with respect to the cooling power of 6×10^{-6} kg s $^{-1}$. The counterflow heat exchanger determines for a large part this lowest T_{SQ} value. The cooling power \dot{q}_{sensor} at the SQUID heat exchanger was measured as a function of the SQUID heat exchanger temperature T'_{SQ} for different mass flows. At the optimal mass flow, $\dot{q}_{sensor} = 0.2$ W at 42 K and roughly 1.2 W at 77 K. A temperature fluctuation of 0.05 K was measured at $T'_{SQ} = 54$ K and a mass flow of 3×10^{-5} kg s $^{-1}$, which is roughly a factor 40 above the required temperature stability. This requirement with respect to the temperature stability can be relaxed when the earth's magnetic field is compensated by means of a coil.

Further research will be performed on this cooling system. In the meantime, the experience gained with this large cooling system can already be used in the design of a smaller configuration cooling system, using miniature Stirling cryocoolers¹⁷.

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